PLEASE CHECK WEB S THE LATI	SITE AT WWW.ONBOARDS ST REVISION OF THIS MA	YSTEMS.CO INUAL	M FOR		
ROTORCR	AA APPROVED AFT FLIGHT MAI SUPPLEMENT	NUAL			
E-69	STC SH1262NW E-69 Load Weigh System for the Airbus Helicopters AS350 Series				
<b>R</b> /N	S/N				
FAA Approved: FAA Approved: Digitally signed by ROBERT Y SCHLEIN Date: 2020.10.19 13:53:24 -07'00' Manager, Northwest Flight Test Section, AIR-715 Federal Aviation Administration Seattle, WA Date: 19 October 2020					
	RFM Supplement	Document I 121-02			
SYSTEMS	Load Weigh System	Page 1 of 10	Revision 3		

Rev.	Date	Page(s)	Reason for Revision
0	August 13, 2004	All	Initial Release.
1	June 14, 2006	All	Revised to new format for the AS350 series and also to match the Eurocopter format (this included adding a separate section titled Emergency Procedures). Added additional system description in Section I, added section IV.1.
2	November 6, 2013	All	Added B3 model, standardized wording with other RFMSs.
3	19 October 2020	All	Added C-40 Indicator model.



<b>RFM Supplement</b>	Document Number 121-024-00		Rev. 3
Load Weigh System	Page	FAA Approv	<sup>/ed</sup>
	2 of 10	19 Octo	ber 2020

## 1 GENERAL

This supplement must be attached to the appropriate FAA approved Airbus Helicopters' Rotorcraft Flight Manual when an Onboard Systems P/N 200-295-00 Load Weigh System is installed in accordance with Supplemental Type Certificate (STC) NO. SH1262NW. The information contained herein supplements or supersedes the basic manual only in those areas listed herein. For limitations, procedures and performance information not contained in this supplement consult the basic Rotorcraft Flight Manual and Airbus Helicopters' Rotorcraft Flight Manual Supplement – External Load Transport.

The Load Weigh System is a compliment to the helicopter lifting system. Its purpose is to display the weight of the load carried on the cargo hook.





# 2 LIMITATIONS

The basic Rotorcraft Flight Manual and Rotorcraft Flight Manual Supplement – External Load Transport issued by Airbus Helicopters remain applicable.

#### Maximum Load

The maximum load to be carried with the Load Weigh System installed is the lesser of that specified by the Airbus Helicopters Flight Manual Supplement – External Load Transport or 3,000 lbs. (1,361 kg).

#### **Placards**

The following placards are included with the Load Weigh System.

Mounted adjacent to the C-39 indicator (not applicable to C-40 model):

TURN THE WEIGHING SYSTEM OFF WHEN NAVIGATION EQUIPMENT IN USE. NO AIRCRAFT OPERATION SHOULD BE PREDICATED ON THE READING OF THE ONBOARD WEIGHING SYSTEM.

Mounted adjacent to both the load weigh system power switch (if installed) and circuit breaker:

ELECTRONIC WEIGHING SYSTEM

# **3 EMERGENCY PROCEDURES**

The basic Rotorcraft Flight Manual and Rotorcraft Flight Manual Supplement – External Load Transport issued by Airbus Helicopters remain applicable.

ONBOARD SYSTEMS	<b>RFM Supplement</b>	Document Number 121-024-00		Rev. 3
	Load Weigh System	Page 5 of 10	FAA Approv 19 Octo	<sup>ved</sup> ber 2020

### 4. NORMAL PROCEDURES <u>Pre-Flight Check</u>

Prior to a flight involving external load operations with the Load Weigh System perform the following procedures.

- 1. Visually check the load cell connector for damage and security.
- 2. Swing the load cell (and cargo hook) to its full extreme positions to verify that it does not reach the limit of its electrical harness range of motion.
- 3. The Load Indicator powers on when the SLING system is armed. Procedures vary depending on the Indicator model installed. Refer to the following.

**For the C-39 model.** After a brief self-diagnostic routine is complete verify the indicator display indicates "0" as shown below (with no load on the cargo hook).

Figure 4.1 C-39 Indicator Display







### 4. NORMAL PROCEDURES continued Pre-Flight Check continued

To zero (or tare) the weight of the long line, net, remote hook, etc. from the displayed load, apply that weight to the cargo hook and press the knob once and the display should zero out. Press the knob twice to un-zero (un-tare) the display and add this weight back in.



The analog bar <u>always</u> displays the unzeroed load. If there is a discrepancy between the analog bar and the displayed load, a large amount of load has likely been zeroed.

	RFM Supplement	Document Number 121-024-00		Rev. 3
	Load Weigh System	Page F 9 of 10	FAA Approv 19 Octol	<sup>ved</sup> ber 2020

## 5. **PERFORMANCE**

The basic Rotorcraft Flight Manual and Rotorcraft Flight Manual Supplement – External Load Transport issued by Airbus Helicopters remain applicable.

The Load Weigh System is intended as a means of MONITORING the weight of the load suspended from the Cargo Hook.

Before lifting a load, it is recommended that the load weight be estimated, the shape/size is considered and, upon lifting the load, monitor the load indicator and compare the actual engine torque value vs. the expected value for a given weight to verify sufficient performance.

UNBOARD SYSTEMS	RFM Supplement	Document Number 121-024-00		Rev. 3
	Load Weigh System	Page FAA Approv 10 of 10 19 Octob		ber 2020