Please check web the late	site at www.onboards est revision of this mai	ystems. nual	com for	
FAA APPROVED ROTORCRAFT FLIGHT MANUAL SUPPLEMENT Load Weigh System for the Sikorsky S-61L and S-61N Model Helicopters				
R/N FAA Approved:Ma	S/N 	cation Off	 ice	
Date: Revised: 15 June 2011				
	RFM Supplement		ent Number -056-00	
SYSTEMS	Load Weigh System	Page 1 of 5	Revision 1	

GENERAL

This supplement must be attached to the appropriate FAA approved Rotorcraft Flight Manual when an Onboard Systems 200-243-00 or 200-243-01 Load Weigh System is installed in accordance with Supplemental Type Certificate (STC) No. SR00681SE.

The information contained herein supplements or supersedes the basic Rotorcraft Flight Manual only in those areas listed herein. For limitations, procedures and performance information not contained in this supplement consult the basic Rotorcraft Flight Manual and Rotorcraft Flight Manual Supplement for the cargo sling assembly

The Load Weigh System is a compliment to the helicopter lifting system. Its purpose is to display the weight of the load carried on the cargo hook. The Load Weigh System consists of three primary components: the cockpit mounted load indicator, the internal electrical harness and the load cell. The load cell is installed between the rotorcraft cable suspension and the cargo hook (not included with this kit).

ENBOARD SYSTEMS	RFM Supplement	Document Number 121-056-00		Rev. 1
	Load Weigh System	Page FAA AI 2 of 5 JUL		oproved 1 5 2011

SECTION 1 - LIMITATIONS

Weight Limits

The maximum allowable load is 12,000 lbs (5,443 kgs) or as specified by the Rotorcraft Flight Manual, whichever is <u>less</u>. The load weigh system's 12,000 lb load rating does <u>NOT</u> supersede the maximum allowable external load as specified by the Rotorcraft Flight Manual

Operation

The basic Rotorcraft Flight Manual and Rotorcraft Flight Manual Supplement for the cargo sling assembly remain applicable.

The load weigh indicator shall be operated in accordance with Section 3 of Owner's Manual 120-082-00.

Placards

Mounted adjacent to the Onboard Systems load indicator in full view of the pilot and co-pilot:

TURN THE WEIGHING SYSTEM OFF WHEN NAVIGATION EQUIPMENT IN USE. NO AIRCRAFT OPERATION SHOULD BE PREDICATED ON THE READING OF THE ONBOARD WEIGHING SYSTEM

Mounted adjacent to both the power switch and circuit breaker in full view of the pilot and co-pilot:

ELECTRONIC WEIGHING SYSTEM

ONBOARD SYSTEMS	RFM Supplement	Document Number 121-056-00		Rev. 1
	Load Weigh System	Page FAA Ap 3 of 5 JUL		pproved 1 5 2011

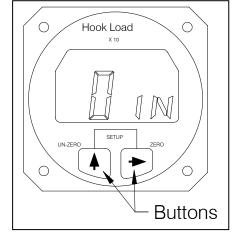
SECTION 2 - NORMAL PROCEDURES

Pre-Flight Check

Prior to a flight involving external load operations perform the following procedures. If the procedures are not successful do not use the equipment until the problem has been corrected.

- 1. Visually check the load cell electrical connections and electrical harness for damage and security.
- 2. Visually check the load cell fasteners for presence and security.
- 3. Initialize the Load Indicator per the following:
 - Power on the Load Indicator and allow it to warm up for 5 minutes (with no load on the hook). Press both Indicator buttons at the same time to go to the setup mode. Scroll through the menu, using the left button, until "0 in" (see Figure 1) is displayed, then press the right button. Remove any weight from the cargo hook that is not to be zeroed out and press either button to complete the procedure.

Figure 1 Load Indicator





SECTION 3 - PERFORMANCE

The basic Rotorcraft Flight Manual and Rotorcraft Flight Manual Supplement for the cargo sling assembly remain applicable.

The Load Weigh System is designed and installed as a means of monitoring the load (weight) suspended from the cargo hook. Functional and performance characteristics have not been determined on the basis of the load cell indication or display. Therefore, this instrument shall <u>NOT</u> be used as a primary indication of performance and flight operation must <u>NOT</u> be predicated on its use.

ENBOARD SYSTEMS	RFM Supplement	Document Number 121-056-00		Rev. 1
	Load Weigh System	Page FAA A _I 5 of 5 JUL		oproved 1 5 2011