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FAA APPROVED

***ROTORCRAFT FLIGHT MANUAL
SUPPLEMENT***

***Onboard Systems
External Load Suspension System
with Keeperless Cargo Hook***

Robinson R22 Series

R/N _____ S/N _____

FAA Approved: *Shun Rippe*
for Manager, Seattle Aircraft Certification Office

Date: *13 JAN 2012*

Revised:



RFM Supplement

Document Number
121-004-01

Cargo Hook Kit

Page
1 of 15

Rev. 0

INTRODUCTION

This supplement must be attached to the appropriate FAA approved Rotorcraft Flight Manual when an Onboard Systems 200-262-01 Cargo Hook Kit is installed in accordance with Supplemental Type Certificate (STC) NO. SR00920SE. The information contained herein supplements or supersedes the basic manual only in those areas listed herein. For limitations, procedures and performance information not contained in this supplement, consult the basic Rotorcraft Flight Manual.

I. LIMITATIONS

I.1 Type of Operation

The basic Flight Manual remains applicable. With a load attached to the cargo hook, operation shall be conducted in accordance with the respective national operational requirements. For U.S. operators FAR Part 133 is applicable.

This cargo hook kit is approved for non-human cargo, class B rotorcraft load combinations.

The helicopter may also be operated with the provisions portion of the kit installed only. This includes the hardpoint, stowed manual and electric release cables and all Cargo Hook related equipment in the cockpit.

	RFM Supplement	Document Number 121-004-01	Rev. 0
	Cargo Hook Kit	Page 2 of 15	FAA Approved JAN 13 2012

I.2 Weight and CG

The maximum weight and CG of the combined helicopter and external load remains the same as the basic manual.

Table I.2-1 Weight and CG data

DESCRIPTION	WEIGHT lbs (kgs)	FUSELAGE STATION in. (mm)	LATERAL STATION in. (mm)
Cargo Attach Point	-	92.2 (2342)	-3.0 (76)
Complete Cargo Hook Kit	5.0 (2.3)	92.2 (2342)	-3.0 (76)
Provisions Kit (no hook)	2.0 (0.9)	92.2 (2342)	-3.0 (76)

Center of gravity limits must be checked with and without the external load to verify that the rotorcraft is within the approved weight and center of gravity limits.



RFM Supplement

Document Number
121-004-01

Rev. 0

Cargo Hook Kit

Page
3 of 15

FAA Approved
JAN 13 2012

I.3 Cargo Hook Load

Maximum Cargo Hook loading is 400 Lbs (181 kgs).

I.4 Airspeed

Vne = 102 KIAS, with Cargo Hook installed, but no load.

Vne = 75 KIAS, or less with external load.

Do not exceed Vne of basic helicopter.



Airspeed with external cargo is limited by controllability. Caution should be exercised when carrying external cargo, as the handling characteristics may be affected by the size, weight, and shape of the cargo load.

It is the operator's responsibility to establish the maximum operational speed for each specific configuration.



RFM Supplement

Document Number
121-004-01


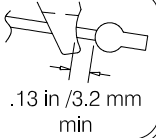
Rev. 0

Cargo Hook Kit

Page
4 of 15

FAA Approved
JAN 13 2012

I.4 Placards, continued

PLACARD	LOCATION
<div style="border: 1px solid black; padding: 5px; text-align: center;">CARGO RELEASE</div>	Mounted adjacent to the cyclic release switch in clear view of the pilot.
<div style="border: 1px solid black; padding: 5px; text-align: center;">CARGO RELEASE</div>	Mounted adjacent to the copilot's release switch in clear view of the pilot.
<div style="border: 1px solid black; padding: 5px; text-align: center;">CARGO RELEASE</div>	Mounted adjacent to the mechanical release in clear view of the pilot.
<div style="border: 1px solid black; padding: 5px; text-align: center;">PULL</div>	Mounted adjacent to the mechanical release in clear view of the pilot.
<div style="border: 1px solid black; padding: 5px; text-align: center;">CARGO</div>	Mounted adjacent to the Cargo Hook circuit breaker in clear view of the pilot.
<div style="border: 1px solid black; padding: 5px;"> <div style="display: flex; align-items: center;"> <div style="text-align: center; flex: 1;">  <p>WARNING</p> <p>Inadvertent loss of load can result from improper cable adjustment. See manual for complete instructions.</p> </div> <div style="flex: 1; text-align: center;">  <p>.13 in / 3.2 mm min</p> </div> </div> </div>	Located on the bottom side of the cargo hook.



RFM Supplement

Document Number
121-004-01

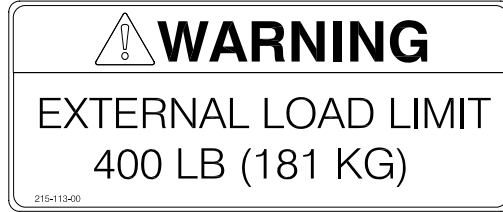
Rev. 0

Cargo Hook Kit

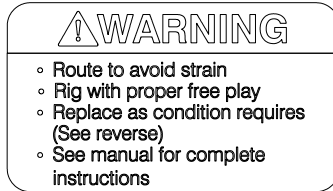
Page
5 of 15

FAA Approved
JAN 13 2012

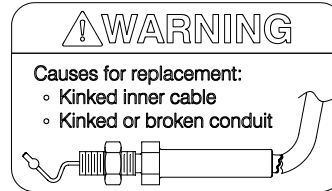
I.4 Placards, continued



Mounted on the belly of the aircraft adjacent to the cargo hook attachment point in clear view of the ground support personnel.

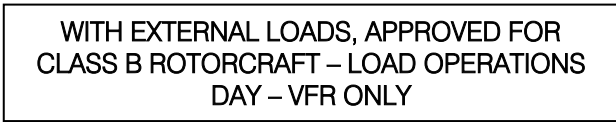


One Side

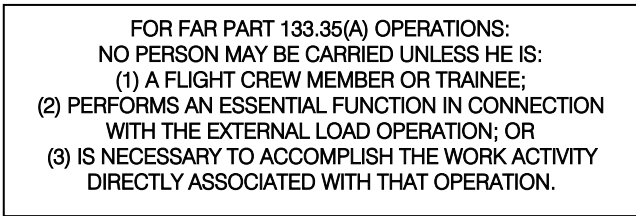


Opposite Side

Located on the manual release cable, near the cargo hook.



Mounted on the instrument panel in clear view of the pilot.



Mounted on the instrument panel in clear view of the pilot.



RFM Supplement

Document Number
121-004-01

Rev. 0

Cargo Hook Kit

Page
6 of 15

FAA Approved
JAN 13 2012

II. NORMAL PROCEDURES

II.1 STATIC DISCHARGE

Instruct the ground crew to ensure that the helicopter has been electrically grounded prior to attaching cargo to discharge static electricity. If possible, maintain ground contact until hook up is completed.

II.2 PRE-FLIGHT CHECK

Before a flight involving external load operations perform the following procedures. If the procedures are not successful do not use the equipment until the problem has been corrected.

1. Visually check all mounting fasteners to ensure that they are tight.
2. Visually check the electrical connector for loose or damaged pins and sockets.
3. Swing the hook and the suspension assembly to their full extremes to verify that they do not reach the limit of the mechanical release cable range of motion and actuate the mechanical release mechanism.
4. Visually check the cargo hook case and covers for cracks and damage.
5. Visually check the load beam for gouges and cracks.



RFM Supplement

Document Number
121-004-01

Rev. 0

Cargo Hook Kit

Page
7 of 15

FAA Approved
JAN 13 2012

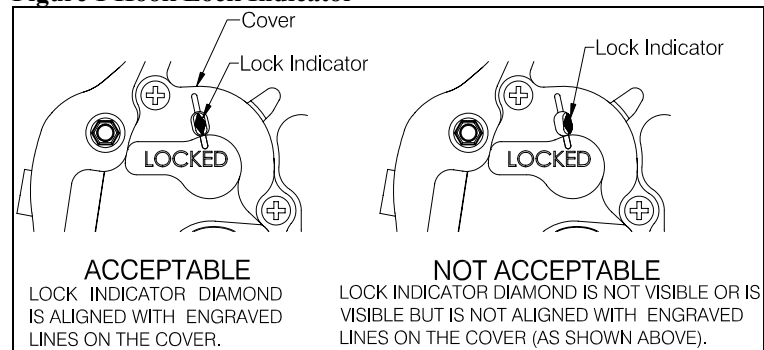
II.2 PRE-FLIGHT CHECK continued

6. Pull the manual release handle in the cockpit to ensure proper operation. The cargo hook load beam must open. Return the cargo hook load beam to the locked position by manually pushing up on it. The load beam should snap shut. Verify that the hook lock indicator on the side of the hook returns to the fully locked position. The cargo hook may be flown in the open position to facilitate loading by a ground crew.



In the closed and fully locked position the hook lock indicator must align with the lines on the manual release cover (see Figure 1).

Figure 1 Hook Lock Indicator



RFM Supplement

Document Number
121-004-01

Rev. 0

Cargo Hook Kit

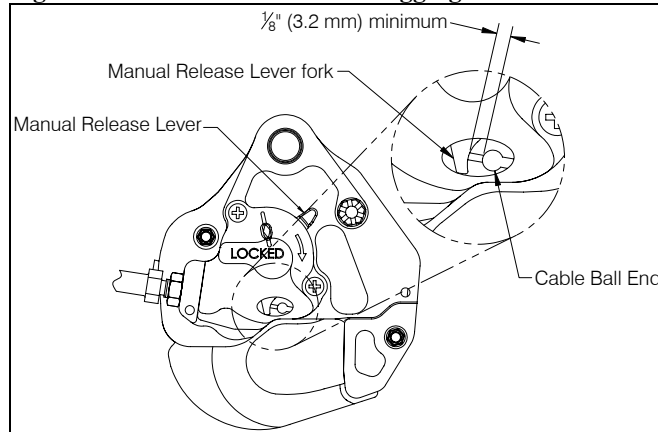
Page
8 of 15

FAA Approved
JAN 13 2012

II.2 PRE-FLIGHT CHECK continued

7. Cycle the cargo hook's electrical release mechanism to ensure proper operation. Pressing the CARGO RELEASE switch on cyclic should cause the cargo hook load beam to open. The cargo hook may be returned to the locked position by manually pushing up on the load beam. The load beam should snap shut. Verify that the hook lock indicator on the side of the hook returns to the fully locked position (see Figure 1).
8. Check the manual release cable rigging through the window in the cargo hook manual release cover. With the cargo hook load beam closed and locked, rotate the manual release lever clockwise to remove the free play (the free play is taken up when the hook lock indicator begins to move, this is also readily felt as the lever rotates relatively easily for several degrees as the free play is taken up) and hold it in this position while checking the gap between the release lever fork and the cable ball end as shown below. Visually check that there is approximately a minimum gap of 1/8" (3.2 mm) as shown in Figure 2.

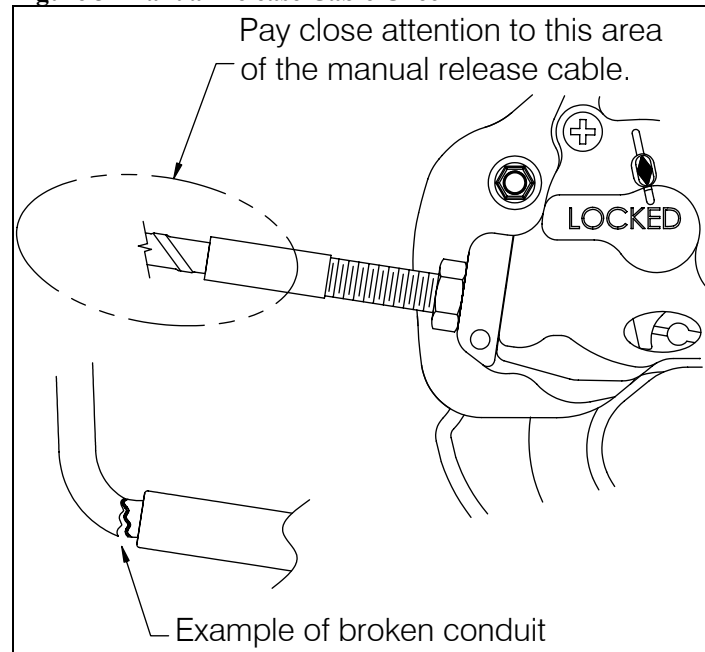
Figure 2 Manual Release Cable Rigging



II.2 PRE-FLIGHT CHECK continued

8. Visually check the manual release cable for damage, paying close attention to the flexible conduit at the area of transition to the cargo hook end fitting (refer to Figure 3). Check for kinked, broken, or splitting of the outer black conduit in this area and separation of the conduit from the steel end fitting. This type of damage is cause for immediate replacement.

Figure 3 Manual Release Cable Check



II.3 CARGO HOOK RIGGING

Extreme care must be exercised in rigging a load to the Cargo Hook. The following illustration shows the recommended rigging configuration.



The example shown is not intended to represent all possibilities. It is the responsibility of the operator to assure the hook will function properly with the rigging.

Some combinations of small primary rings and large secondary rings could cause fouling during release.



RFM Supplement

Document Number
121-004-01

Rev. 0

Cargo Hook Kit

Page
11 of 15

FAA Approved
JAN 13 2012

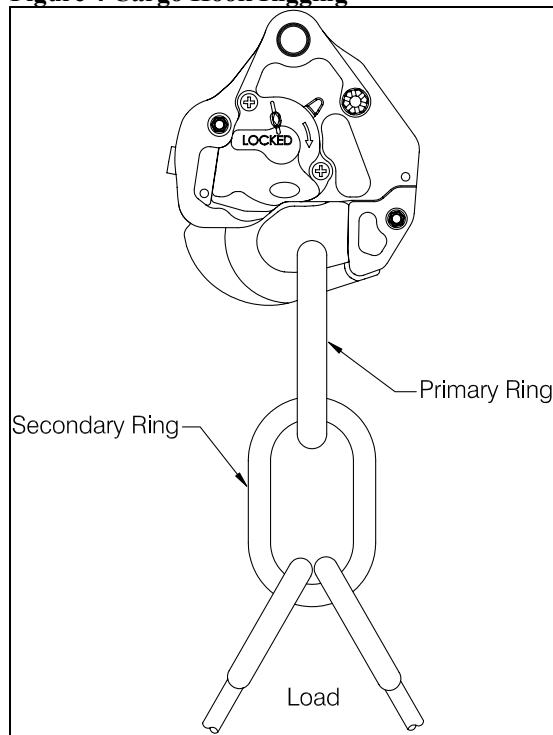
II.3 CARGO HOOK RIGGING, continued

Nylon Type Straps or Rope



Nylon type straps (or similar material) or rope must not be used directly on the cargo hook load beam. If nylon straps or rope must be used they should be first attached to a steel primary ring. Verify that the ring will freely slide off the load beam when it is opened. Only the primary ring should be in contact with the cargo hook load beam. See Figure 4.

Figure 4 Cargo Hook Rigging



RFM Supplement

Document Number
121-004-01

Rev. 0

Cargo Hook Kit

Page
12 of 15

FAA Approved
JAN 13 2012

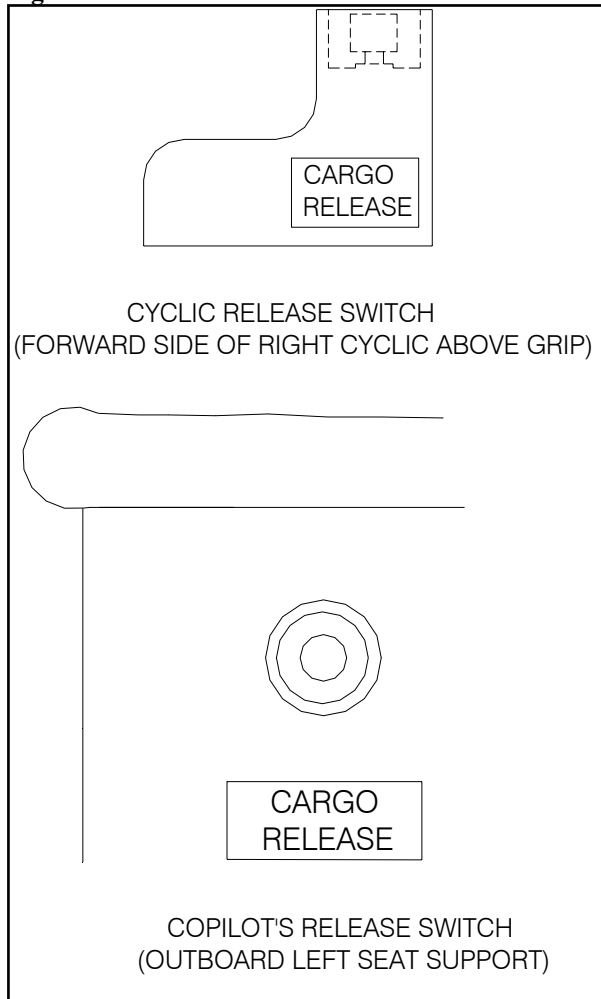
II.4 IN-FLIGHT OPERATION

Note

Control movement should be made smoothly and kept to a minimum to prevent oscillation of the load.

Actuate either electrical release switch to release the external load.

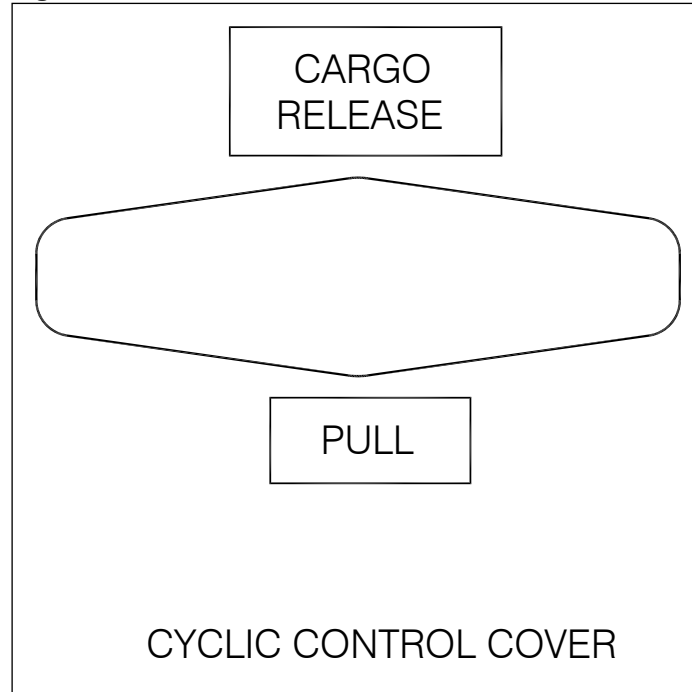
Figure 5 Electrical Release Switches



II.4 IN-FLIGHT OPERATION, continued

The mechanical release handle may be used to release the external load in normal circumstances. Pull up on it to release the load.

Figure 6 Mechanical Release Handle



It is the responsibility of the operator to establish safe operational limits for each specific configuration.

	RFM Supplement	Document Number 121-004-01	Rev. 0
	Cargo Hook Kit	Page 14 of 15	FAA Approved JAN 13 2012

III. EMERGENCY PROCEDURES

III.1 CARGO FAILS TO RELEASE ELECTRICALLY

In the event that the Cargo Hook will not release the external load electrically, proceed as follows:

1. Maintain tension on the sling.
2. Pull the mechanical release handle to release the external load.

IV. PERFORMANCE

The basic Flight Manual issued by Robinson Helicopter Company remains applicable. There is no change from basic flight performance with no load attached to the Cargo Hook. Performance will be reduced depending on the size, weight and shape of the external load.



RFM Supplement

Document Number
121-004-01

Rev. 0

Cargo Hook Kit

Page
15 of 15

FAA Approved
JAN 13 2012