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***E-69 Electronic
Load Weigh System
For The
Eurocopter AS350 Series***

Owner's Manual

*Owner's Manual Number 120-022-00
Revision 9
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RECORD OF REVISIONS

<i>Revision</i>	<i>Date</i>	<i>Page(s)</i>	<i>Reason for Revision</i>
4	11/01/00	ii, 1-1, 4-1, deleted Section 6	Removed reference to E-70 and moved it to new manual 120-102-00.
5	9/17/02	Title, 4-3	Factory address change.
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7	10/09/07	TOC, Section 1, 2-3, 2-6, 3-7 to 3-9, 3-12, 4-1 & 4-2	Added explanation of warnings, cautions and notes to section 1. Updated warnings, cautions and notes throughout. Changed “daily inspection to “daily check.”
8	3/2/10	4-1 & 4-2	Changed overhaul frequency schedule.
9	03/02/18	4-1 through 4-5	Updated 5 year/1000 hour inspection actions. particle inspection of load cell, removed daily check, updated definition of “hours of external load operations” . Updated table 4-1

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CONTENTS

***Section 1* General Information**

Introduction, **1-1**
Safety Labels, **1-1**
System Part Number, **1-2**
Indicator Features, **1-2**
Indicator Specifications, **1-2**
Indicator Pin Out, **1-3**
Load Cell Specifications, **1-3**
Bill of Materials, **1-4**
Inspection, **1-4**

***Section 2* Installation Instructions**

Introduction, **2-1**
System Installation Overview, **2-1**
Internal Harness Installation, **2-1**
Indicator Installation, **2-2**
Indicator Internal Back Light, **2-2**
Indicator Hook-Open Warning, **2-2**
Remote Analog Meter, **2-2**
350 “Cargo Sling” Load Cell Installation, **2-3**
350 “Cargo Swing” Load Cell Installation, **2-3**
Electrical Connections, **2-6**
Installation Check-Out, **2-7**
Weight and Balance, **2-7**
Paper Work, **2-7**

***Section 3* Operation Instructions**

Indicator Front Panel, **3-1**
The Run Mode, **3-2**
 To Zero or Tare the Display, **3-3**
 To Un-Zero the Display, **3-3**
 Error Codes, **3-4**
The Setup Mode, **3-5**
 Indicator Dampening, **3-7**
 To look at or change the dampening level, **3-7**
 Indicator Calibration, **3-8**
 To look at or change the calibration code, **3-8**

***Section 3 (cont.)* Operation Instructions**

Installation Zero, **3-9**

To run the installation zero routine, **3-9**

Calibration by Lifting a Known Weight, **3-9**

To run calibration by a known weight routine, **3-10**

Setting the Scale on a Remote Analog Meter, **3-11**

To look at or change the scale, **3-11**

Select KG or LB Units, **3-12**

To look at or change the units, **3-12**

Indicator Version, **3-13**

***Section 4* Maintenance**

Lubrication, **4-1**

Inspection, **4-2**

Trouble Shooting **4-4**

Instructions for Returning a System to the Factory, **4-5**

***Section 5* E-69 Certification**

STC, **5-1**

STA, **5-3**

Flight Manual Supplement, **5-5**

Figures

- 2-1 350 Sling System Installation Overview, **2-4**
- 2-2 350 Swing System Installation Overview, **2-5**
- 2-3 Wiring Arrangement, **2-6**
- 3-1 Front Panel, **3-1**
- 3-2 After Diagnostic Routine, **3-2**
- 3-3 LB Legend Displayed, **3-2**
- 3-4 Zeroing the Display, **3-3**
- 3-5 Changing Dampening Level, **3-7**
- 3-6 Changing the CAL Code, **3-8**
- 3-7 Running CAL Routine, **3-10**
- 3-8 Entering Load in CAL Routine, **3-10**
- 3-9 Changing the Scale, **3-11**
- 3-10 Changing the Units, **3-12**
- 3-11 Looking at Indicator Version, **3-13**

Tables

- 1-1 Indicator Specifications, **1-2**
- 1-2 Indicator Pin Out, **1-2**
- 1-3 Load Cell Specifications, **1-3**
- 3-1 Indicator Error Codes, **3-4**
- 3-2 Indicator Setup Routines, **3-6**
- 4-1 Inspection Criteria, **4-3**
- 4-2 Trouble Shooting, **4-4**

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Section 1

General Information

Introduction

The Onboard Load Weigh system is a compliment to the helicopter lifting system. Its purpose is to display the weight of the load carried on the cargo hook. The Load Weigh System consists of three components, the cockpit mounted Indicator, the Internal Harness and the Load Cell. The system is designed specifically for each helicopter and is intended to be a permanent installation.

Safety Labels

The following definitions apply to safety labels used in this manual.



Indicates a hazardous situation which, if not avoided, will result in death or serious injury.



Indicates a hazardous situation which, if not avoided, could result in death or serious injury.



Indicates a hazardous situation which, if not avoided, could result in minor or moderate injury.



Draws the reader's attention to important or unusual information not directly related to safety.



Used to address practices not related to personal injury.

System Part Number

Part Number	Description
200-058-00	E-69 Load Weigh System

Indicator Features

The features of the C-39 Indicator include:

- Front panel programmable
- Data Recorder communications link
- Internal back lighting system
- Night Vision Compatible (NVG) When an external NVG source is used in place of the internal back light.
- Hook-Open Warning Display
- Analog Meter Output Signal

Indicator Specifications

Table 1-1 Indicator Specifications

SPECIFICATIONS	210-095-00 INDICATOR
Size	Fits standard 2¼" clock hole
Weight	7 oz
Operating Voltage	16 to 31 VDC
Current Consumption	< 25 mA
Accuracy Over Operating Temperature Range	0.1% ± 1 digit
Operating Temperature Range	+70°C to -45°C
Storage Temperature Range	+80°C to -50°C
Scaleable Analog Output	0 to 5VDC ± 0.5%

Indicator Pin Out

The connector located on the back of the Indicator has the following pin out.

Table 1-2 Indicator Pin Out

Pin Letter	Function
A	+ 28 VDC In
B	- Load Cell Signal
C	+ Load Cell Signal
D	+ Load Cell Excitation
E	Load Cell Common
F	Analog Out Common
G	+ Analog Out
H	Hook Open
J	Data Recorder Clock
K	Data Recorder Data
L	Shield
M	Back Light Common
N	Back Light Source 28 VDC
P	Aircraft Ground
R	Not Used

Load Cell Specifications

Table 1-3 Load Cell Specifications

SPECIFICATIONS	LOAD CELL
Weight	1.5 pounds
Accuracy Over Operating Temperature Range	0.5% ± 1 digit
Operating Temperature Range	+70°C to -45°C
Storage Temperature Range	+80°C to -50°C

Bill Of Materials

The following items are included with each system, if shortages are found contact the distributor from whom the system was purchased.

Part Number	Description	Mfg.	Mfg. P/N	Qty
120-022-00	Owner's Manual	Onboard	Same	1
210-095-00	C-39 Indicator Assembly	Onboard	Same	1
210-046-00	E-69/E-70 Load Cell Assy	Onboard	Same	1
270-048-01	Harness Assembly	Onboard	Same	1
400-048-00	Power Switch	C&K	7101-SYZQE	1
215-010-00	Placard	Onboard	Same	2
215-012-00	Placard	Onboard	Same	1
512-001-00	Ty-Wrap	All-States	RT-350-NA	10
510-028-00	Screw	Any	AN515-4R10	6
510-029-00	Nut	Any	AN365-440A	6
510-062-00	Washer	Any	AN 960-4	10
235-035-00	QD Bracket	Onboard	Same	1

Inspection

Inspect each component for evidence of mishandling and damage. If damage is evident, do not use it. File a claim with the carrier and notify the distributor from whom the system was purchased.

Section 2

Installation Instructions

Introduction

This section describes how to install the components of the Electronic Load Weigh System.

System Installation Overview

Refer to the load cell installation sections for an overview of the various system installations.

Internal Harness Installation

The Internal Harness is made up of four cables terminated to one connector. The connector is plugged into the back of the Indicator. One of the cables is marked "LOAD CELL" and is fitted with a bulkhead fitting. This cable is connected to the load cell. Another cable is marked "POWER" and is connected to the aircraft electrical power. Another cable is marked "LIGHT", refer to the *Indicator Internal Back Light* section for installation instructions. The last cable is marked "DATA" and is connected to the optional Data Recorder or Analog Slave Meter.

The load cell cable can be routed with the hook manual and electrical cables to the cargo hook area. The load cell cable can be attached to the hook manual and electrical disconnect bracket using the furnished 235-035-00 bracket and hardware. The bracket location should be close enough to the load cell to insure the load cell cable is not stressed when the cargo hook is moved to its furthestmost point, but far enough away to minimize excess cable which may be snagged.

Route the harness to the electrical Buss and to the Indicator mounting location. Secure the cables to the existing wiring bundles with the Ty-wraps. If it is necessary to remove the load cell bulkhead connector to ease cable routing, reconnect using the color code below.

<u>Wire Color</u>	<u>Connector Pin</u>
Red	A
Green	B
White	C
Black	D
Shield	E

Secure the cables clear of flight control rods.

Indicator Installation

The Indicator should be mounted in a position that is convenient, accessible and visible to the pilot. It can be mounted in a standard 2¼" instrument hole. Connect the Indicator to its Internal Harness, refer to *Internal Harness Installation*.

Indicator Internal Back Light

The 210-095-00 Indicator is equipped with an Internal Back Lighting System that can be connected to the aircraft 28 VDC light dimming circuit. Use a 22 gauge, twisted pair, shielded cable to connect the aircraft dimming circuit to the Internal Harness. Connect the cable shield wire to airframe ground at the light dimmer end of the cable **ONLY**.

Indicator Hook-Open Warning

The 210-095-00 Indicator is equipped with a Hook-Open Warning feature that can be connected to a cargo hook equipped with a hook open switch. Depending on the capabilities of the cargo hook switch, the Indicator will flash "HOOK OPEN" when the cargo hook load beam is open. The cargo hook switch must be normally open when the cargo hook load beam is in the closed position. When the load beam is open, one side of the switch must be grounded and the other side of the switch is to be connected to the Indicator. Use a 22 gauge, shielded wire to connect the cargo hook switch to the Indicator. Disassemble the Indicator mating connector and carefully solder the wire, from the cargo hook switch, to pin H. Connect the cable shield wire to airframe ground as close to the cargo hook as possible, at the cargo hook end of the cable **ONLY**.

Remote Analog Meter

The 210-095-00 Indicator is equipped with an Analog drive circuit that can be connected to a remote analog meter. Use a 22 gauge, twisted pair, shielded cable to connect the Remote Analog Meter to the Indicator. Disassemble the Indicator mating connector and carefully solder the positive wire, from the analog meter, to pin G and the common wire to pin F. Connect the cable shield wire to airframe ground as close to the Analog Meter as possible, at the Analog Meter end of the cable **ONLY**.

The 210-095-00 Indicator can be connected to Onboard Systems' Analog Slave Meter, P/N 210-180-00, through the "DATA" cable. This meter gives solid weight indications without needle bounce. The Analog Slave Meter may be mounted in any convenient location in a standard 3" instrument hole. Use Cable, P/N 270-059-00 to connect the Analog Slave Meter to the Internal Harness "DATA" cable.

350 “Cargo Sling” Load Cell Installation

Separate the cargo hook from the aircraft, at the cargo hook. Attach the load cell lower lifting eye to the cargo hook using the existing hardware. Note, the load cell cable should point to the left.

Attach the load cell upper lifting eye to the aircraft lifting gimbal using the existing hardware.



If the aircraft is equipped with a French electronic load cell, it will be necessary to purchase several items from Aerospatiale to complete the installation.

Description	Part Number	Quantity
Gimbal	350A86-1026-00	1
Bolt	350A86-1027-21	1
Nut	22453BC100L	1

Swing the hook assembly to the full extremes to verify that it does not stress the cables or self trip.

350 “Cargo Swing” Load Cell Installation

Separate the cargo hook from the aircraft, at the cargo hook. Attach the load cell lower lifting eye to the cargo hook using the existing hardware. Note, the load cell cable should point to the left.

Attach the load cell upper lifting eye to the aircraft lifting gimbal using the existing hardware.

Secure the load cell cable with ty-wraps, such that it will not chafe on the cargo swing.

Swing the cargo hook to its full extremes to verify that it does not stress the cables or self trip.

Figure 2-1 350 Sling System Installation Overview

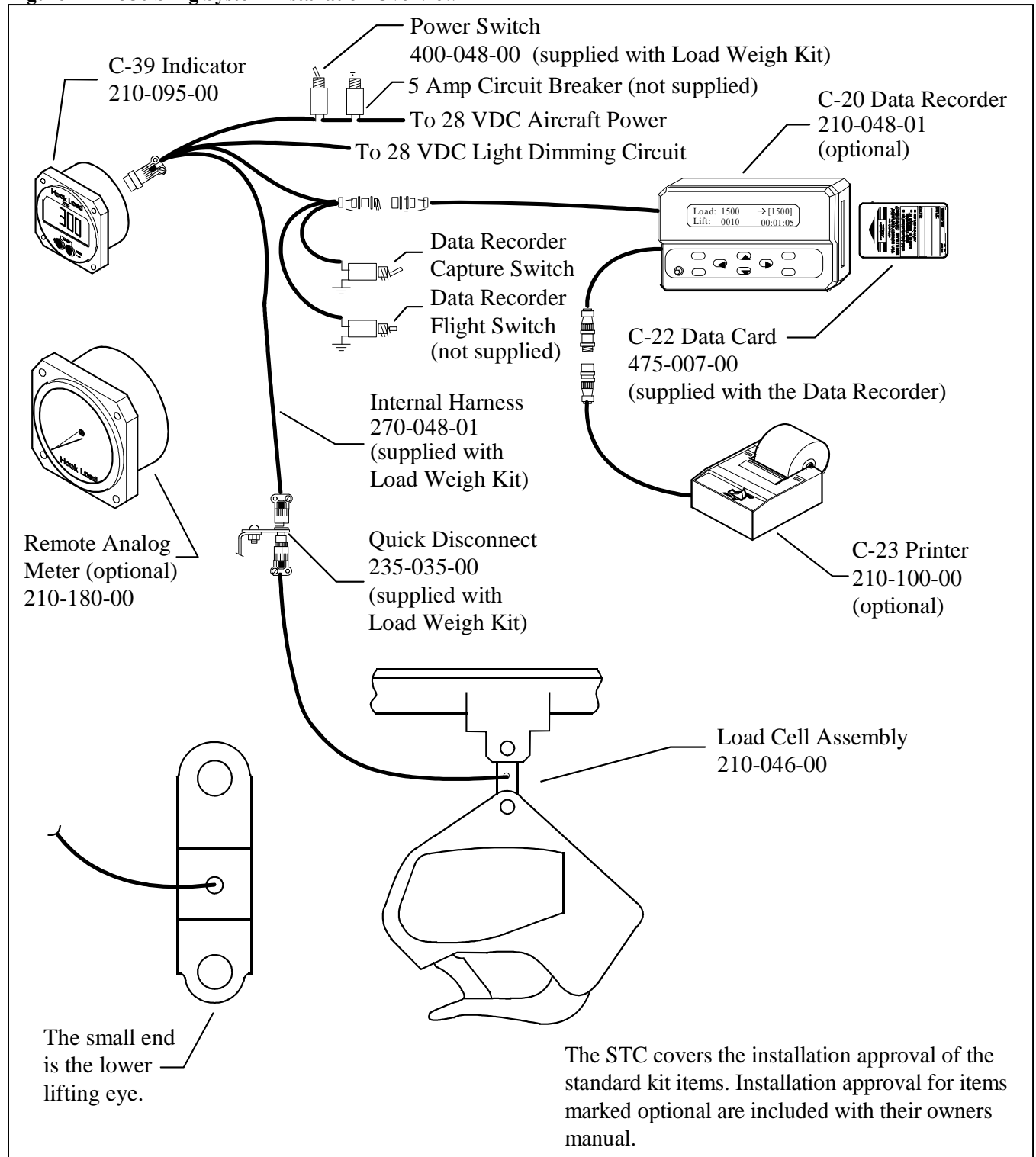
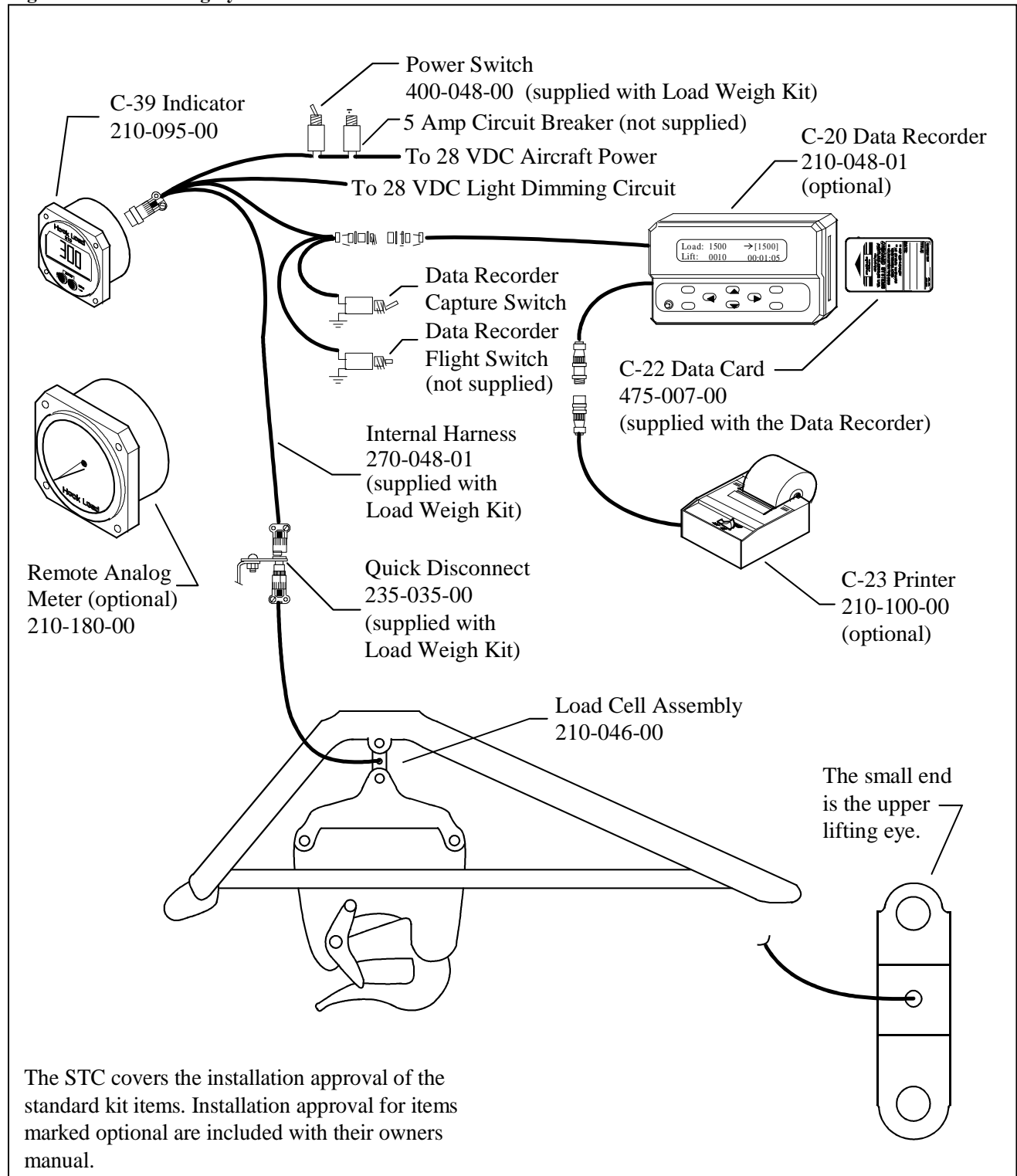


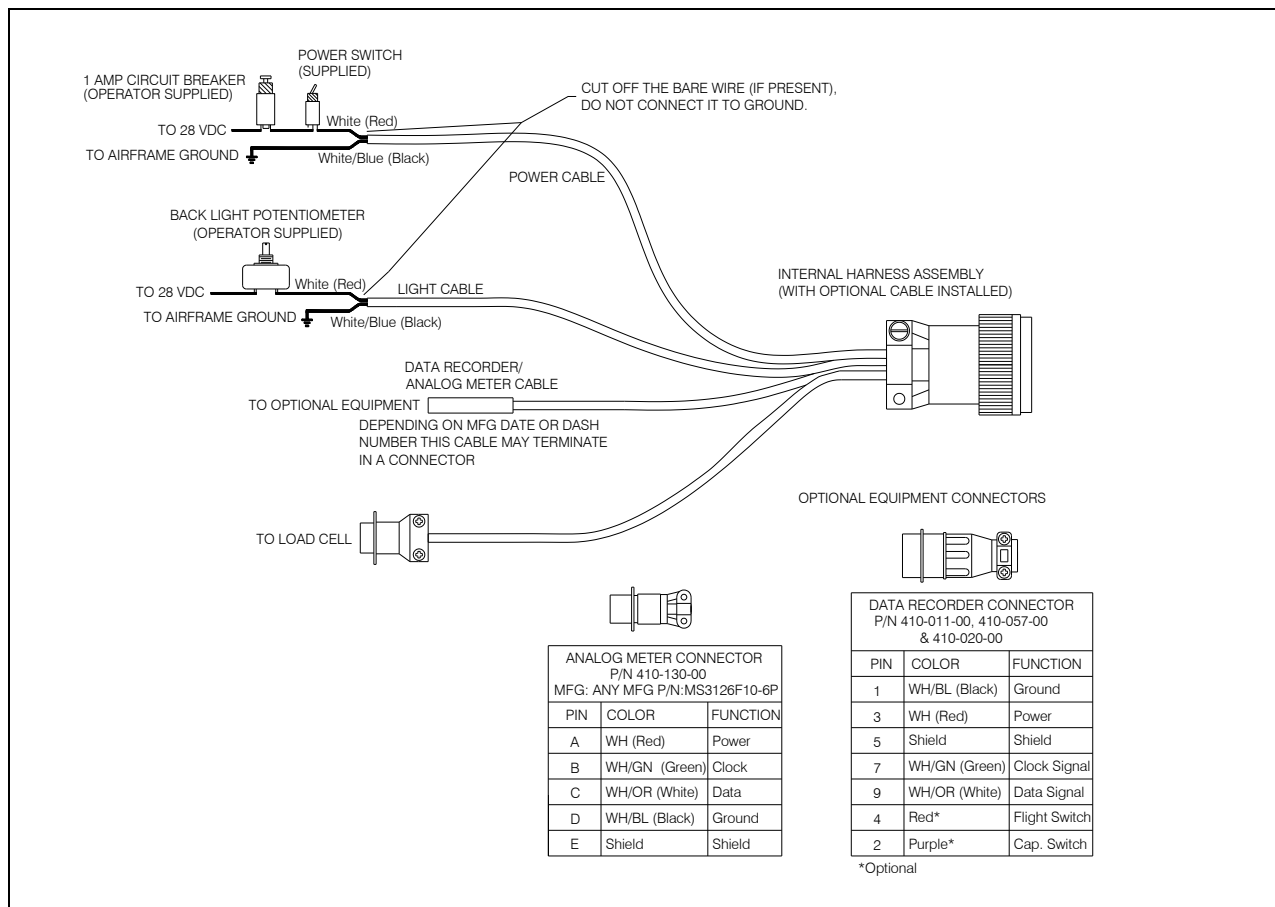
Figure 2-2 350 Swing System Installation Overview



Electrical Connections

Install the supplied power switch, part number 400-048-00. The “POWER” cable on the Internal Harness is supplied extra long, cut off the excess cable and use as needed to connect the switch and circuit breaker. Connect the “POWER” red wire to one side of the power switch, connect another piece of suitable red wire to the other side of the switch and then to an available 1 or 2 amp circuit breaker as illustrated in Figure 2-3. Connect the circuit breaker to the 24 VDC buss. Connect the black wire to the ground buss. The bare wire should be cut off as it is not needed at this end of the cable. Use a minimum of 22 gauge wire to make all connections. Secure the connections and protect from corrosion.

Figure 2-3 Wiring Arrangement



Connect the Internal Harness to the Indicator connector. Install the placard 215-010-00 “ELECTRONIC WEIGHING SYSTEM” next to the power switch and circuit breaker. Install the placard 215-012-00 “TURN THE WEIGHING SYSTEM OFF WHEN NAVIGATION EQUIPMENT IN USE” “NO AIRCRAFT OPERATION SHOULD BE PREDICATED ON THE READING OF THE ONBOARD WEIGHING SYSTEM” next to the Indicator.

NOTICE

If the C-23 Printer is being utilized with the C-20 Data Recorder, a 5 amp circuit breaker should be used.

Installation Check-Out

After the system has been properly installed, activate the circuit breaker to turn the system on. Refer to *Operation Instructions*.

The installers authorized inspector (AI) must accomplish both a satisfactory ground and flight EMI test for each installation. (Ref AC: 25-10 paragraph 5(m) and 5(n) Special emphasis is needed for those helicopters equipped with electronic fuel control, electronic stability augmentation systems, electronic flight controls, or any other electronic device which could affect the rotor craft safety.

Ensure that the cargo hook is free to move to its full extremes.

Ensure that all electrical cables are secured clear of flight control rods and hydraulic lines.

Weight and Balance

ITEM	WEIGHT
Indicator	.43 pounds
Load Cell	1.5 pounds
Total	1.93 pounds

Paper Work

Insert the Flight Supplement into the aircraft flight manual. In the US fill in FAA form 337 for the initial installation. This procedure may vary in different countries. Make the appropriate aircraft log book entry.

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Section 3

Operation Instructions

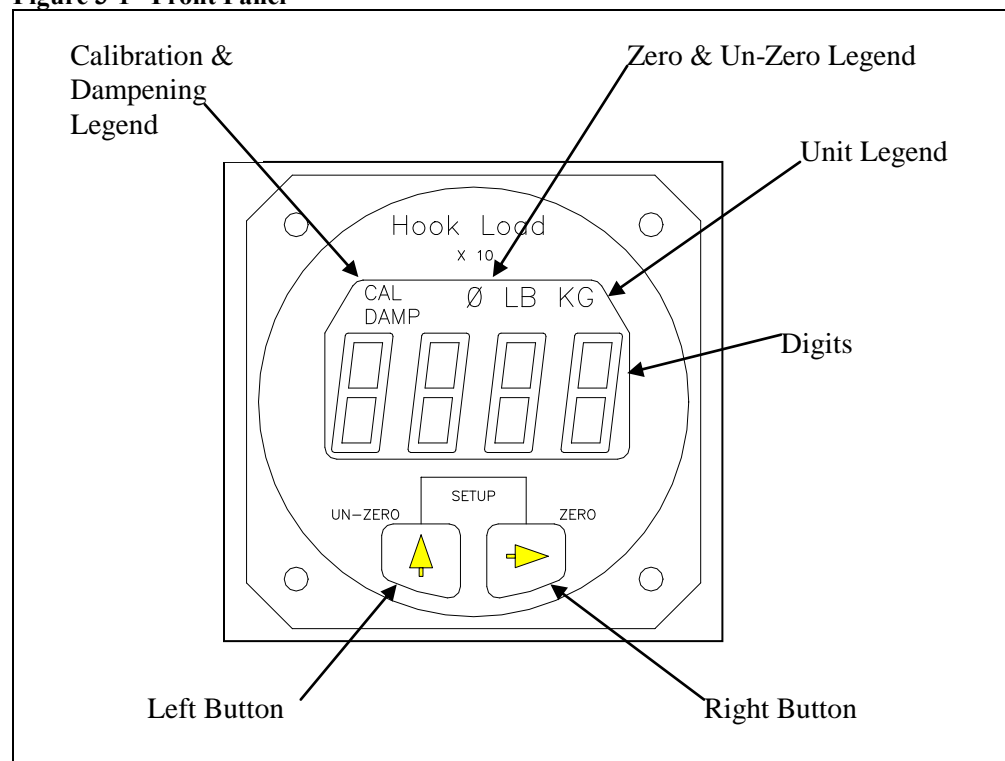
Load Weigh System

Indicator Front Panel

The C-39 Indicator front panel includes the following features.

- The four 7 segment LCD digits show the weight on the Cargo Hook and displays various Setup information.
- The Legends clarify the digital display. i.e. when the LB Legend is turned on, the display will be pounds, etc.
- The Right button is used to Zero the display in the Run Mode and select the digit to be changed in the Setup Mode.
- The Left button is used to Un-Zero the display in the Run Mode and scroll the selected digit in the Setup Mode.

Figure 3-1 Front Panel

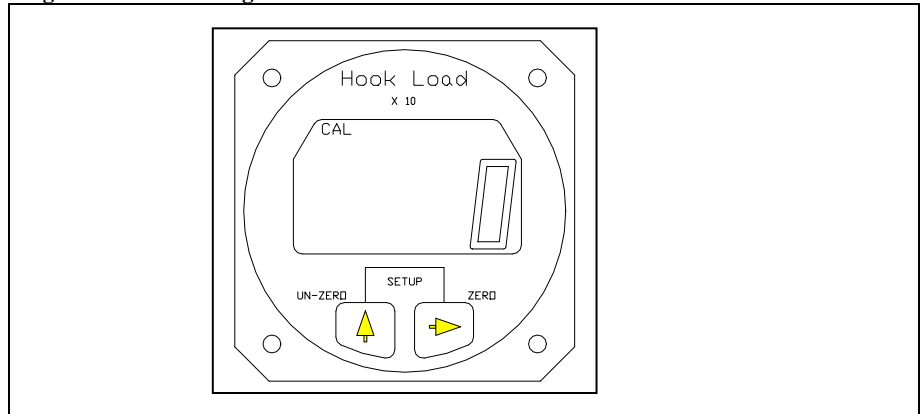


The Run Mode

The C-39 Indicator has two operating modes, Run and Setup. The Run Mode is used to display the cargo hook weight and the Setup Mode is used to setup or configure the Indicator to the helicopter and to the Load Cell. When powered up, the Indicator always comes to life in the Run Mode.

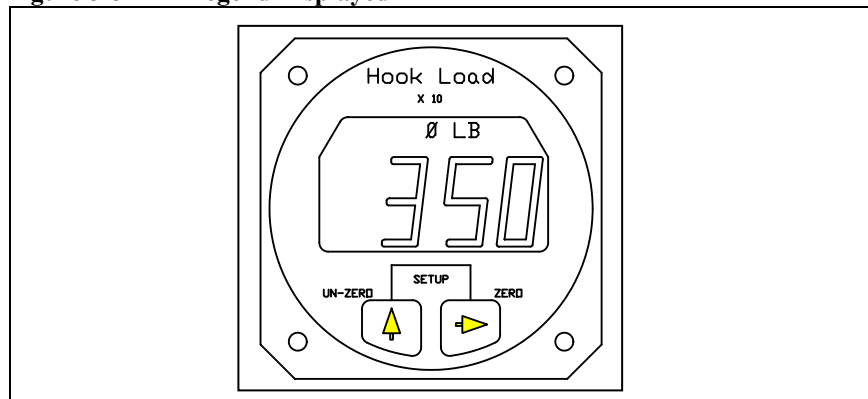
After the Indicator has been correctly installed, power it up by activating the Load Weigh Circuit Breaker. The Indicator will go through a self diagnostic routine. During this routine the display will display all of the digits and legends. If a problem is found during the routine an Error Code will be displayed. For an explanation of Error Codes see the section *Error Codes*. After the diagnostic routine the display should look like this:

Figure 3-2 After Diagnostic Routine



The illustration is of the Indicator in the Run Mode with no load on the hook. Note the LB legend displayed.

Figure 3-3 LB Legend Displayed

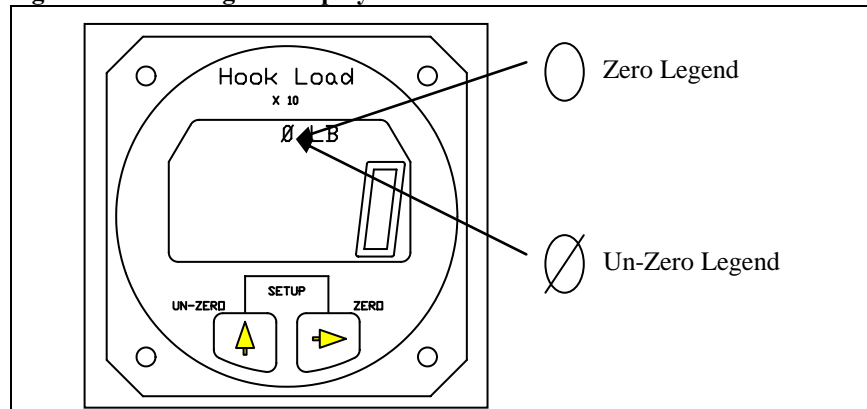


The illustration is a typical hook load reading. The display is 3,500 pounds, note the last digit is not displayed.

To Zero or Tare the Display

The zero feature is used to zero or tare the weight on the Cargo Hook that is not wanted, such as the weight of a cargo net or long line. The Right button is used to zero the Indicator reading. When the Right button is pressed the display is zeroed. The zero legend is turned on and the zeroed number is stored in memory. If the Right button is pressed again, before the Un-zero button is pressed, the display blinks in response to the button closure. Zero is only available in the Run Mode.

Figure 3-4 Zeroing the Display



To Un-Zero the Display

The Left button is used to add the zeroed value back into the current Indicator reading or Un-zero the display. When the Left button is pressed, the number previously zeroed is added to the current display and the Un-zero legend is turned on. If the Left button is again pressed before the zero button is pressed, the display blinks in response to the button closure. Un-Zero is only available in the Run Mode.

Error Codes

Error Codes are the result of difficulties discovered during the Indicator diagnostic tests. Diagnostic tests occur at power up and during the execution of certain routines. Listed below is a matrix of the Error Code displays, their meaning and possible corrective action. Pressing either button will usually bypass the error code, however, the displayed information may be suspect.

Table 3-1 Indicator Error Codes

DISPLAY	CAUSE	POSSIBLE CORRECTIVE ACTION
Err 1	A/D or D/A circuit failure	Potential short in the optional analog meter cable. Clear short and power cycle the Indicator by turning the power to the Indicator off for a few moments. If Error Code continues, return the Indicator to the factory.
Err 2	NV Ram failure	Power cycle the Indicator; if Error Code continues, return the Indicator to the factory.
Err 3	NV Ram write failure	Re-enter data, if Error Code continues, return the Indicator to the factory.
Err 4	NV Ram busy failure	Power cycle the Indicator, if Error Code continues return the Indicator to the factory.

The Setup Mode

The C-39 Indicator can be used with a wide range of helicopters and load cells. The Setup Mode on the Indicator matches the Indicator to the Load Cell and to the helicopter. This is done by entering data into the Indicator. Entered data includes the load cell Calibration Code, the units that the Indicator should read-out (pounds or kilograms), and several other items.

The Indicator has a group of Setup routines, arranged in menu form, that are used to configure the Indicator. Shown on the next page is a matrix of the Setup routines and a brief discussion of their function and how they are programmed. A complete discussion of each setup item is presented later in this section.

To enter the Setup Mode press both the Right and Left buttons at the same time while the Indicator is powered up and in the Run Mode. To exit the Setup Mode and return to the Run Mode, press both the buttons at the same time. If you are in a Setup routine and have started to change an entry, but you change your mind before completing the procedure, power cycle the Indicator to exit the Setup Mode and then go to the Run Mode without changing the item. The Indicator is power cycled by turning the Indicator power off for a few moments.

The Setup Mode, continued

Table 3-2 Indicator Setup Routines

MENU	FUNCTION	DISPLAY
Press the Left button to scroll through the menu	Press the Right button to view or change the menu item.	To return to the Run Mode press both the Right and Left buttons at the same time.
DAMP	<u>Dampening Level</u> , sets the pilots preference for display dampening.	Blinking display is previously entered Dampening Level. Select the desired dampening level by pressing the Left button.
CODE	<u>Calibration Code</u> , matches the Indicator to the Load Cell.	Display is previously entered CAL Code. The Code is changed by selecting the digit to be changed with the Right button. The selected digit will blink. Change the blinking digit by pressing the Left button.
0 in	<u>Installation ZERO</u> , matches the Indicator to the installed Load Cell and to the helicopter. After this procedure the display will be zero when no load is on the Cargo Hook.	Display is a combination of load on the Load Cell, and normal load cell zero offset. Remove all weight from the installed Load Cell except the Cargo Hook, and press any button to complete the procedure and return to the Run Mode.
LOAD	<u>Load</u> , is used to calibrate the system by lifting a known load.	No previous display is shown. Enter the known load using the Right button to select the digit to be changed and Left button to enter the number. Known load is entered "X 10" i.e.; 5000 kilograms is entered as 500. After the known load is entered, press both buttons at the same time and lift the known load. When the load is stabilized press either button. A new Calibration Code will be calculated and the known load will be displayed. This completes the procedure.
Scale	<u>Scale</u> , matches the analog output of the Indicator to an optional remote analog meter.	Display is previously entered number. To change the number use the Right button to select a digit, use the Left button to scroll the digit to the desired number. Entry is times 10.
LB KG	<u>Units</u> , selects the Indicator units (pounds or kilograms).	Display is previously selected unit. To change the unit, use the Left button.
XX - V	<u>Version</u> , is the revision level of the Indicator hardware and software.	Version is for information only, it cannot be changed.

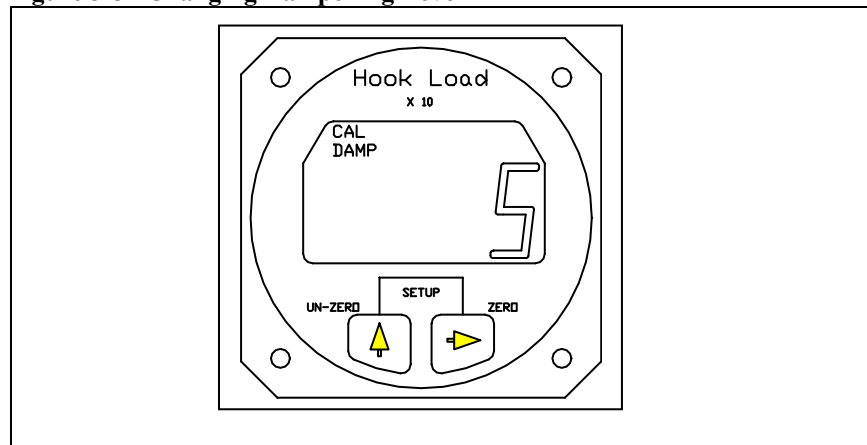
Indicator Dampening

The Damp or dampening routine allows the pilot to adjust the Indicator dampening level to his preference. The dampening routine is a program that stabilizes the Indicator reading. It offers a trade-off between Indicator responsiveness and stability. Ten dampening levels are available, from 0 through 9. At level 0 the display responds to the slightest change in weight. However, if the load bounced even slightly, the display digits would respond instantly, making the display look unstable. With a dampening level of 9, the display would be stable under the most turbulent conditions, however, it would take several seconds for the display to respond to a change in weight. The ideal dampening level will depend on the flying conditions. A mid range setting of 5 or 6 is usually adequate.

To Look at or Change the Dampening Level

With the Indicator powered up and in the Run Mode, press both buttons at the same time to go to Setup. Scroll through the menu, using the Left button, until the word DAMP is displayed. To look at or change the Dampening Level press the Right button. The display should look like this:

Figure 3-5 Changing Dampening Level



The CAL and the DAMP legend is turned on and the previously set dampening level is displayed. To return to Run without changing the current dampening level press both the Right and Left buttons at the same time. To change the dampening number, use the Left button to scroll the blinking digit to the desired number. After the selection has been made press both the Right and Left buttons at the same time to return to Run.

Indicator Calibration

The Calibration Code, or CAL code, is a mandatory input. The Indicator will not accurately display the load without the correct Calibration Code. The Calibration Code scales the signal from the Load Cell.

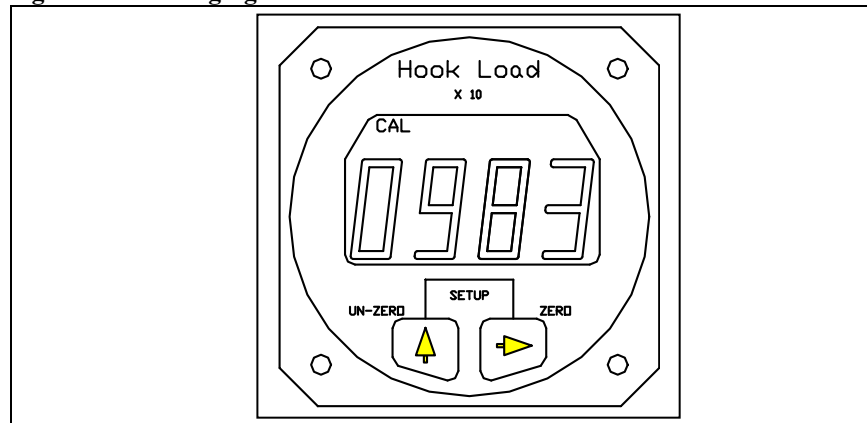
Indicator Calibration, continued

If the C-39 Indicator was supplied as part of a Load Weigh System, the Calibration Code will have been entered into the Indicator by the factory, however, it should be confirmed. If the Indicator is to be mated to a different Load Cell, it must be calibrated before use. Calibration can be done by entering a known Calibration Code or by lifting a known load and having the Indicator calibrate itself. Both options are discussed below.

To Look at or Change the Calibration Code

With the Indicator powered up and in the Run Mode, press both buttons at the same time to go to Setup. Scroll through the menu until the word CODE is displayed, then press the Right button. The display should look like this:

Figure 3-6 Changing the CAL Code



The CAL legend is turned on and the previously entered or computed Calibration Code is displayed. To return to Run without changing the CAL Code, press both the Right and Left buttons at the same time. To change the Calibration Code, use the Right button to select the digit to be changed, then use the Left button to scroll the blinking digit to the desired number. When the Calibration Code has been entered, press both the Right and Left button at the same time to return to Run.

NOTICE

Depending on the type of Load Cell, the Calibration code could be a 3 or 4 digit number. If the Calibration Code is a 3 digit number a leading zero (0) must be used. For example if a Load Cell had a CAL Code of 395 it would be entered as 0395.

If the load cell Calibration Code is not known or as a cross check, the Indicator can generate the Calibration Code. This is done by entering the weight of a known load into the Indicator LOAD routine and then lifting the load. See the section *Calibration by Lifting a Known Load*.

Installation Zero

Installation zero is a routine that matches the Indicator to the ***INSTALLED*** Load Cell. It adjusts the Indicator reading to compensate for the weight of the Cargo Hook on the Load Cell and whatever zero offset is built into the Load Cell. The Installation Zero procedure is not mandatory. If done the Indicator will read zero when the Un-Zero button is pressed and there is no weight on the Cargo Hook. If the Installation Zero is not done, the Indicator will show the weight of the Cargo Hook plus the value of the Load Cell zero offset.

To Run the Installation Zero Routine

With the Indicator powered up and in the Run Mode, press both buttons at the same time to go to Setup. Scroll through the menu until the symbol "0 in" is displayed, then press the Right button. The CAL legend will be turned on and the current weight on the Cargo Hook will be displayed and blinking. Remove any weight that is not to be zeroed out and press either button to complete the procedure and return to the Run Mode.

Calibration by Lifting a Known Weight

Calibration by lifting a known weight is a Setup routine that calculates the Calibration Code for the Load Cell attached to the Indicator. It is useful if the load cell Calibration Code is not known or as a cross check to the accuracy of a known Calibration Code. The procedure is done by entering the known weight into the Indicator and then lifting the weight. This procedure can be done in the shop or on the helicopter. The accuracy of the procedure is directly related to the weight of the known load. If for example the procedure was done with a 1,000 pound load that was assumed to weigh only 900 pounds, all subsequent lifts would be displayed 10% light.



Be sure to include the weight of everything between the Cargo Hook and the load, i.e. the cable, net, dirt, etc.

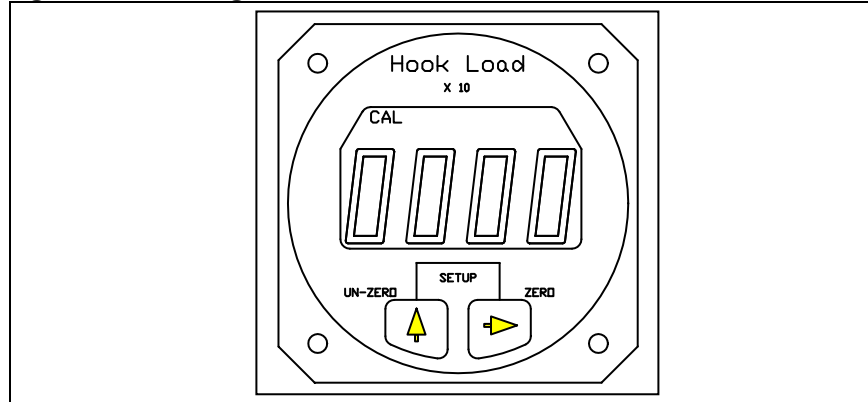
The closer the known load approaches the lifting capacity of the helicopter, the more accurate the calculated Calibration Code will be.

Calibration by Lifting a Known Weight, continued

To Run the Calibration by Lifting a Known Weight Routine

With the Indicator powered up and in the Run Mode, press both buttons at the same time to go to Setup. Scroll through the menu until the word LOAD is displayed, then press the Right button. The display should look like this:

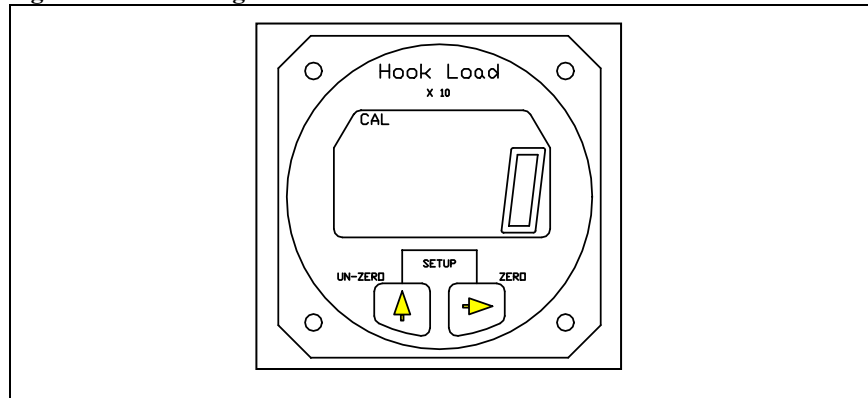
Figure 3-7 Running CAL Routine



The CAL legend is turned on and the first digit is blinking. The previous load is not displayed. At this point if you wish to return to the Run Mode without changing the Calibration Code, power cycle the Indicator. At this point it is not possible to return to the Run Mode without changing the Calibration Code by using the buttons on the Indicator front panel.

To proceed with the procedure, use the Right button to select the digit to be changed, then use the Left button to scroll the blinking digit to the desired number. Note that the known weight is entered "X 10"; a 1000 pound load is entered as 100. When the known load has been entered, press both the Right and Left button at the same time. The display will look like this:

Figure 3-8 Entering Load in CAL Routine



Calibration by Lifting a Known Weight, continued

The CAL legend and the digits will be blinking. Again, at this point if you wish to return to the Run Mode without changing the Calibration Code, power cycle the Indicator. It is not possible to return to the Run Mode by using the buttons on the Indicator front panel without changing the Calibration Code. If you wish to proceed, lift the known load and when it is stabilized, press either button to complete the procedure. The Indicator will display the load. This ends the procedure. The Indicator is now calibrated to the Load Cell. It is a good practice to go to the Code routine and record the new Calibration code for later reference.

Setting the Scale for a remote analog meter

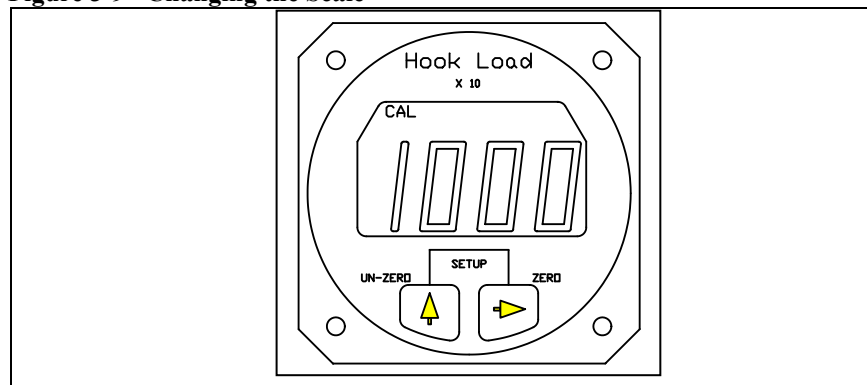
The Scale routine is used when a user supplied analog meter is connected to the Indicator. It is used to match or calibrate the analog meter to the Indicator. The Indicator outputs a 0 to 5 VDC analog signal which is proportional to the Load Cell load. The Scale number tells the Indicator at what point in pounds or kilograms it should reach the 5 VDC output. If for example a 5 volt analog meter is used and its full scale reading is 10,000 pounds, the number entered into the Indicator Scale routine would be 1000 (the number is entered X 10). This number tells the Indicator that it should output the proportional 0 to 5 VDC signal between zero pounds and 10,000 pounds.

The Scale number does not affect Onboard Slave Meters, P/N 210-106-00 or 210-180-00. This number only affects user supplied instruments connected to the analog out signal.

To Look at or Change the Scale

With the Indicator powered up and in the Run Mode, press both buttons at the same time to go to Setup. Scroll through the menu until the word SCALE is displayed, then press the Right button. The display should look like this:

Figure 3-9 Changing the Scale



To Look at or Change the Scale, continued

The CAL legend is turned on and the previously set Scale number is displayed. To return to Run without changing the Scale, press both the Right and Left button at the same time. To change the Scale number, use the Right button to select a digit to be changed, then use the Left button to scroll the blinking digit to the desired number. When the complete Scale number has been entered, press both the Right and Left button at the same time to return to Run.

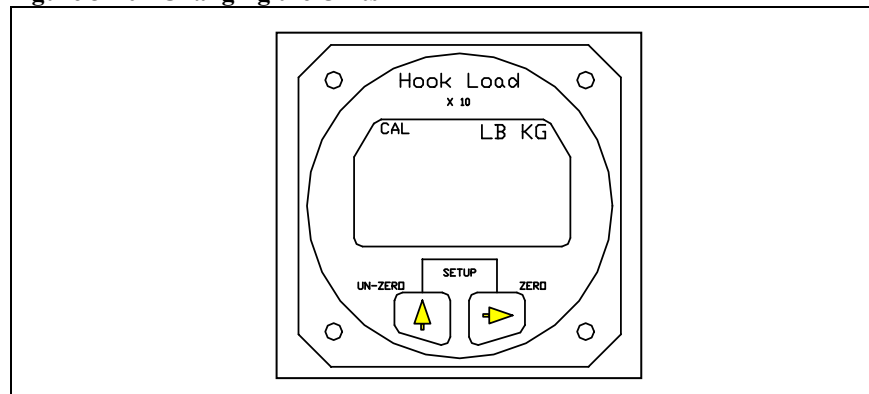
Select KG or LB Units

The units routine sets the display to read in pounds (LB) or kilograms (KG).

To look at or change the Units

With the Indicator powered up and in the Run Mode, press both buttons at the same time to go to Setup. Scroll through the menu until the word LB or KG is displayed, then press the Right button. The display should look like this:

Figure 3-10 Changing the Units



The CAL legend is turned on and the previously set unit is displayed. To return to Run without changing the units, press both the Right and Left button at the same time. To change the units press the Left button. When the selection has been made, press both the Right and Left button at the same time to return to Run.

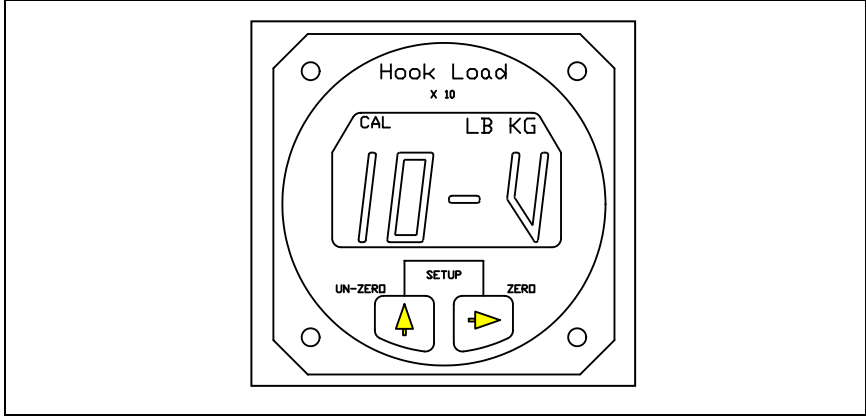
NOTICE

The selected units are displayed when in the Run Mode.

Indicator Version

The Version routine displays the Indicator's hardware and software revision levels. Version is set at the factory and cannot be changed.

Figure 3-11 Looking at Indicator Version



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Section 4

Maintenance

The following procedures are provided for the benefit of experienced aircraft maintenance facilities capable of carrying out the procedures. They must not be attempted by those lacking the necessary expertise. It is recommended that only minor repairs be attempted by anyone other than the factory.

Lubrication

Lubrication of the Load Weigh System is recommended every 500 hours of operation. To obtain maximum life under severe duty conditions such as logging or seismic work, it is recommended to lubricate the Load Cell Assembly pivot points every 200 hours. Recommended lubricants are AeroShell 17, MIL-G-21164 or Mobilgrease 28, MIL-G-81322.

Inspection

The scheduled inspection intervals noted below are maximums and are not to be exceeded. If the load weigh system is subjected to unusual circumstances, extreme environmental conditions, etc., it is the responsibility of the operator to perform the inspections more frequently to ensure proper operation.

Annually or 100 hours of external load operations, whichever comes first, inspect the load weigh system per the following instructions (see Table 4-1 for inspection criteria).

NOTICE

*Hours of external load operations should be interpreted to be (1) anything is attached to the primary cargo hook (whether or not a useful load is being transported) and (2) the aircraft is flying. If these conditions are **NOT** met, time does **NOT** need to be tracked.*

1. Move the load cell and the cargo hook throughout their full ranges of motion and observe the load cell electrical harness to ensure that it has enough slack. The harness must not be the stops that prevent the load cell or cargo hook from moving freely in all directions.
2. Visually inspect the electrical harness strain relief at the load cell for damage.
3. Visually inspect the external load cell electrical harness for damage and chafing.
4. Visually inspect the load weigh harness connector at the belly of the helicopter for damage and security.
5. Visually inspect the load cell covers for corrosion, damage and security (refer to Table 4-1 for limits).
6. Visually inspect the load cell link for corrosion, damage and cracks (refer to Table 4-1 for limits).

Inspection continued

Every 5 years or 1000 hours of external load operations, whichever comes first, perform the following.

1. Return the Load Cell Assembly (P/N 210-046-00) to the factory for inspection and calibration. The factory will inspect the condition of the load cell and perform acceptance test procedures including calibration and zero balance, repairing as necessary.
2. Inspect internal electrical harness from the load weigh indicator to the load cell for general condition, security of attachment, and chafing along the length of wire runs.
3. Inspect for security of load weigh indicator attachment.

Table 4-1 Inspection Criteria

PART	INSPECT FOR:	REPAIR
Load Cell Assembly 210-046-00	Dents, nicks, cracks, gouges, corrosion or scratches in the load cell link.	Repair dents, gouges, nicks, scratches and corrosion if less than .020" deep, blend out at a ratio of 20:1, length to depth, replace assembly if otherwise damaged. Touch up blended areas with zinc chromate primer.
	Dents, nicks, cracks, gouges, corrosion or scratches in the covers.	Repair dents, gouges, nicks, scratches and corrosion if less than .030" deep, blend out at a ratio of 20:1, length to depth. Touch up blended areas with zinc chromate primer.

Trouble Shooting

Table 4-2 Trouble Shooting

PROBABLE CAUSE	DIFFICULTY	CORRECTIVE ACTION
Short in the system, faulty circuit breaker or switch.	Circuit breaker opens when the circuit to Load Weigh System is energized.	Repair or replace defective wiring, circuit breaker and switch.
Faulty wiring, circuit breaker or switch.	Load Weigh Indicator does not light up.	Check the power switch, circuit breaker and wiring. If this doesn't help, return the unit to the factory.
	Where Am I?	Turn the Indicator power off for a few moments. When it comes to life it will be in the Run mode.
Incorrect Calibration Code.	Displayed load is incorrect.	Insure the correct Calibration Code has been entered.
Dampening level is too small.	Displayed load is not stable.	Adjust the Dampening level to a larger number.
Dampening level is too large.	Displayed load takes too long to change the reading when the load is changed.	Adjust the Dampening level to a smaller number.
NV Ram failure, A/D or D/A circuit failure.	Do not recognize the displayed numbers on the Indicator.	Refer to <i>Error Codes</i> in section 3.
Defective load cell or damaged internal harness.	Load Weigh Indicator does not change with changing hook loads.	Check for damaged internal harness, replace load cell.

Instructions for Returning a System to the Factory

If an Onboard Systems product must be returned to the factory for any reason (including returns, service, repairs, overhaul, etc.) obtain an RMA number before shipping your return.



An RMA number is required for all equipment returns.

- To obtain an RMA, please use one of the listed methods.
 - Contact Technical Support by phone or e-mail (Techhelp@OnboardSystems.com).
 - Generate an RMA number at our website: <http://www.onboardsystems.com/rma.php>
- After you have obtained the RMA number, please be sure to:
 - Package the component carefully to ensure safe transit.
 - Write the RMA number on the outside of the box or on the mailing label.
 - Include the RMA number and reason for the return on your purchase or work order.
 - Include your name, address, phone and fax number and email (as applicable).
 - Return the components freight, cartage, insurance and customs prepaid to:

Onboard Systems
13915 NW 3rd Court
Vancouver, Washington 98685
USA
Phone: 360-546-3072

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Section 5

E-69 Certification

STC

United States of America
 Department of Transportation—Federal Aviation Administration
Supplemental Type Certificate

Number SH1262NW

This certificate, issued to **Onboard Systems
 13915 NW 3rd Court
 Vancouver, WA 98685**

*certifies that the change in the type design for the following product with the limitations and conditions therefor as specified hereon meets the airworthiness requirements of Part * of the Federal Aviation Regulation. * FAR 21.29 and FAR 27 effective 2/1/65 plus Amendment 27-1 through 27-10, plus FAA Special Conditions No. 27-79-EU-23, dated 8/13/77.*


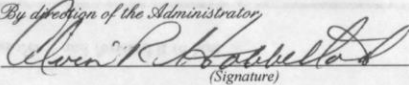
Original Product—Type Certificate Number: H9EU
Make: Eurocopter France
Model: AS-350C, AS-350D, AS-350D1, AS-350B, AS-350B1, AS-350B2, and AS-350BA

Description of the Type Design Change: Fabrication of either Onboard Systems Hydraulic Model H-69 (P/N 200-018-00) or Electronic Models E-69 (P/N's 200-058-00 and 200-295-00) Cargo Hook Load Cell Systems in accordance with FAA Approved Onboard Systems Master Drawing List No. 155-010-00, Revision 10, dated June 15, 2004, or later FAA approved revision, and Installation of either the H-69 system in accordance with FAA approved Onboard Systems Owners Manual No. 120-009-00, Revision "B", dated January 10, 1990, or later FAA approved revision, or the E-69 systems in accordance with FAA approved Onboard Systems Owners Manual No. 120-022-00, Revision 6, dated January 28, 2003, or later FAA approved revision, for the P/N 200-058-00, or FAA approved Onboard System Owners Manual No. 120-114-00, Revision 0, dated June 15, 2004, or later FAA approved revision for the P/N 200-295-00. (See Continuation Sheet)

Limitations and Conditions: Approval of this change in type design applies only to the Eurocopter helicopter models identified above. This approval should not be extended to other aircraft of these models on which other previously approved modifications are incorporated unless it is determined by the installer that the relationship between this change and any of those other previously approved modifications, including changes in type design, will introduce no adverse effect upon the airworthiness of that helicopter. (See Continuation Sheet)

This certificate and the supporting data which is the basis for approval shall remain in effect until surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Administration.

Date of application: May 1, 1981 *Date reissued:* August 2, 1990
Date of issuance: June 23, 1981 *Date amended:* 3/22/88; 8/2/90; 9/20/90; 12/23/91; 3/17/92; 2/12/93; 8/13/04; 3/15/05

 *By design of the Administrator*

 (Signature)
 Acting Manager, Seattle Aircraft Certification Office
 (Title)

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

This certificate may be transferred in accordance with FAR 21.47.

FAA FORM 8110-2(10-68)

United States of America

Department of Transportation—Federal Aviation Administration

Supplemental Type Certificate

(Continuation Sheet)

Number SH1262NW

Onboard Systems

Reissued: August 2, 1990

Amended: March 22, 1988, August 2, 1990, September 20, 1990, December 23, 1991, March 17, 1992, February 12, 1993, August 13, 2004; March 15, 2005

Description of Type Design Change continued:

Inspect the H-69 load cell in accordance with Section 4.6 of Onboard Systems Owners Manual No. 120-009-00, Revision "B", dated January 10, 1990, or later FAA approved revision, and the E-69 load cells in accordance with Section 4 of Onboard Systems Owners Manual No. 120-022-00, Revision 6, dated January 28, 2003, or later FAA approved revision for the P/N 200-058-00 or in accordance with Section 4 of Onboard Systems Owners Manual No. 120-114-00, Revision. 0, dated June 15, 2004, or later FAA approved revision for the P/N 200-295-00.

Limitations and Conditions continued:

Rotorcraft modified in accordance with this STC must be operated in accordance with a copy of one of the following FAA approved Rotorcraft Flight Manual Supplements (RFMS): for Model H-69 (P/N 200-018-00) the RFMS, dated August 2, 1990, in Section 8 of Document 120-009-00, for Model E-69 (P/N 200-058-00) the RFMS, dated February 12, 1993, in Section 5 of Document 120-022-00 and for E-69 (P/N 200-295-00) the RFMS Document 121-024-00, Revision 0, dated August 13, 2004, or later FAA approved revisions. A copy of this Certificate, Continuation Sheet No. SH1262NW the appropriate maintenance manual and appropriate FAA approved Rotorcraft Flight Manual Supplement must be maintained as part of the permanent records of the modified helicopter.

-- END --

Any alteration of this certificate is punishable by a fine of not exceeding \$1,000, or imprisonment not exceeding 3 years, or both.

FAA FORM 8110-2-1 (10-69)

This certificate may be transferred in accordance with FAR 21.47.

PAGE 3 OF 3 PAGES

STA

DEPARTMENT OF TRANSPORT

Supplemental Type Approval

Number: SH97-29

This approval is issued to:

Issue No.: 1

Onboard Systems
11212 NW St. Helens Road
Portland, Oregon
97231

Approval Date: May 15, 1997

Issue Date: May 15, 1997

Responsible Region:

Pacific

Aircraft/Engine Type or Model:

Aerospatiale AS-350C, AS-350D, AS-350D1, AS-350B, AS-350B1, AS350B2 and AS350BA

Canadian Type Approval or Equivalent:

H-83

Description or Type Design Change:

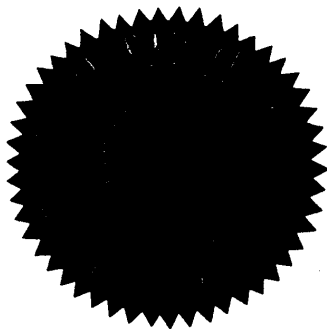
Installation of Onboard Systems Model 2700 Cargo Hook Load Cell System per FAA STC No. SH1262NW

Installation/Operating Data,
Required Equipment
and Limitations:

Fabrication of Onboard Weighing Systems Model 2700 Cargo Hook Load Cell System in accordance with FAA Sealed Onboard Weighing Systems Master Document List No. 2700, Revision 1, dated 09/03/88*; and (see continuation sheet)

Installation of this system in combination with an FAA approved Cargo Swing or Cargo Sling system in accordance with FAA sealed Onboard Weighing Systems Installation Instructions No. 2700, revised 22/03/88*; or (see continuation sheet)

(*or later FAA approved revisions)



Conditions: This approval is only applicable to the type/model of aeronautical product specified therein. Prior to incorporating this modification, the installer shall establish that the interrelationship between this change and any other modification(s) incorporated **will not** adversely affect the airworthiness of this modified product.

L.B. Samoil

Regional Airworthiness Engineer
For Minister of Transport

26-0402 (11-91)

Made from recovered materials

Canada

DEPARTMENT OF TRANSPORT

Supplemental Type Approval

(Continuation Sheet)

Number: SH97-29

Issue No.: 1

Approval Date: May 15, 1997

Issue Date: May 15, 1997

NOTE: THIS ADDENDUM SHALL REMAIN PART OF THE SUPPLEMENTAL TYPE APPROVAL REFERRED THEREIN.

Fabrication of either Onboard Systems Model H-69 (hydraulic) or E-69 (electronic) Cargo Hook Load Cell System in accordance with FAA Approved Onboard Systems Master Drawing List No. 155-010-00, dated 10/08/90*.

Installation of either the H-69 system in accordance with FAA approved Onboard Systems Owners Manual No. 120-009-00, Revision "B", dated 10/01/90*, or the E-69 system in accordance with FAA approved Onboard System Owners Manual No. 120-022-00, dated 10/08/90*.

Inspect the H-69 load cell in accordance with Section 4.6 of Onboard Systems Owners Manual No. 120-009-00, Revision "B", dated 10/01/90*, and the E-69 load cell in accordance with Section 5.3 of Onboard Systems Owners Manual No. 120-022-00, dated 10/08/90*.

Required Equipment and Limitations:

Rotorcraft modified in accordance with this STC must be operated in accordance with a copy of one of the following FAA approved Rotorcraft Flight Manual Supplements: H-69 (hydraulic) revised 02/08/90, or E-69 (electronic) dated 20/09/90*. A copy of this Certificate, Continuation Sheet No. SH1262NW, the appropriate maintenance manual and appropriate FAA approved Rotorcraft Flight Manual Supplement must be maintained as part of the permanent records of the modified helicopter.

(* or later FAA approved revisions)

Flight Manual Supplement

Onboard Systems
11212 NW. St. Helens Rd.
Portland, OR 97231
STC No. SH1262NW

FAA APPROVED

ROTORCRAFT FLIGHT MANUAL SUPPLEMENT

FOR

AEROSPATIALE MODEL AS-350C, AS-350D,

AS-350D1, AS-350B, AS-350B1, AS350B2 and AS350BA HELICOPTERS

R/N _____ S/N _____

This supplement must be attached to the appropriate DGAC approved Aerospatiale Rotorcraft Flight Manual when an Onboard Systems Model E-69 Cargo Hook load cell is installed in accordance with Supplemental Type Certificate (STC) NO. SH1262NW. The information contained herein supplements or supersedes the basic manual only in those areas listed herein. For limitations, procedures, and performance information not contained in this supplement, consult the basic Rotorcraft Flight Manual.

I. LIMITATIONS

Placard - Mount adjacent to the Onboard Systems digital/analog indicator in full view of the pilot and copilot:

TURN THE WEIGHING SYSTEM OFF WHEN NAVIGATION EQUIPMENT IN USE
NO AIRCRAFT OPERATION SHOULD BE PREDICATED ON THE READING OF THE
ONBOARD WEIGHING SYSTEM

- Mount adjacent to both the power switch and circuit breaker in full view of the pilot and copilot:

ELECTRONIC WEIGHING SYSTEM

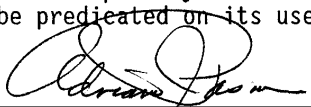
II. PROCEDURES

After installation of this system, swing hook assembly to full extremes to verify that it does not self trip.

III. PERFORMANCE

The hook load weighing system is designed and installed as a means of MONITORING the load (weight) suspended from the cargo hook. Functional and performance characteristics have not been determined on the basis of the load cell indication or display. Therefore, this instrument shall NOT be used as a primary indication of performance and flight operation must NOT be predicated on its use.

FAA Approved:


Manager, Special Certification Branch
Seattle Aircraft Certification Office

Date: September 20, 1990
Revised: March 17, 1992
February 12, 1993

Page 1 of 1