# **European Aviation Safety Agency**



## SUPPLEMENTAL TYPE CERTIFICATE

### EASA.IM.R.S.01136 Revision 1

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 1702/2003 to

## **Onboard Systems International**

13915 NW 3rd Court Vancouver WA 98685 United States

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Product Type Certificate Number: FAA TC H3WE

Type Certificate Holder: McDonnell Douglas

Model: MD 369D, 369E, 369F, 369FE, 369HE,

369HM, 369HS, 500N, 600N

Original STC Number: FAA STC SH4908NM

#### **Description of Design Change:**

Installation of Onboard Systems Model 200-034-00, 200-034-01, and 200-034-02 Load Weigh Kits in accordance with Onboard Systems Master Drawing List No.155-014-00, revision 14, dated August 28, 2008, or later EASA approved revision.

#### **Associated Technical Documentation:**

- Installation: Installation of the Load Weigh Kits must be in accordance with Onboard Systems Owner's Manual No. 120-017-00, Revision 8, dated August 28, 2008, or later EASA approved revision.
- Inspection and Maintenance: This modification must be inspected and maintained in accordance with Section 4 of the Onboard Systems Owner's Manual No. 120-017-00, Revision 8, dated August 28, 2008, or later EASA approved revision.
- Operation: Rotorcraft modified in accordance with STC must be operated in accordance with Onboard Systems Rotorcraft Flight Manual Supplement (RFMS) 121-035-00, Rev. 1, dated 1 October 2008, or later EASA approved revision.

#### **Limitations and Conditions:**

- 1. Prior to installation of this modification the installer must determine that the interrelationship between this modification and any other previously installed modification will introduce no adverse effect upon the airworthiness of the product. The installation of this modification by third persons is subject to written permission of the approval holder and holding and disposal of the approved appropriate documentation.
- 2. Approval of this change in type design applies to only those MD Helicopter model rotorcraft listed above, which were previously equipped with an FAA approved installation of MD Helicopter cargo hook kit and cargo hook listed in the table or an Onboard Systems 200-187-00 or 200-264-00 cargo hook kit.



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Cargo Hook Kit	Helicopter Model	Cargo Hook
369H90072-501, -505, -507, -515, -519	369D	369H92105-501
369H90072-505, -517, -519	369E	369H92105-501
<b>369</b> H90072-505	369F	369H92105-501
369H90072-505, -511, -519	369FF	369H92105-501
369H90072-519, -523	500N	369H92105-501
<b>369</b> H90072-501	369HE	369H92105-501
<b>369H90072-5</b> 01	369HM	369H92105-501
<b>369H90072-5</b> 01	369HS	369H92105-501
<b>369H90072-</b> 525, -529	600N	369H92105-503, -505

This Certificate shall remain valid unless otherwise surrendered or revoked.

For the European Aviation Safety Agency,

Date of issue: 15 July 2009

/ Massimo MAZZOLETTI Certification Manager

STC - EASA.IM.R.S.01136 Revision 1 - Onboard Systems International