SUPPLEMENTAL TYPE CERTIFICATE

10077086

This Certificate/Approval is issued by EASA, acting in accordance with Regulation (EU) 2018/1139 on behalf of the European Union, its Member States and of the European third countries that participate in the activities of EASA under Article 129 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

ONBOARD SYSTEMS INTERNATIONAL

13915 NW 3RD COURT
VANCOUVER WA 98685
UNITED STATES OF AMERICA

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and, if applicable, environmental protection requirements when operated within the conditions and limitations specified below:

Type Certificate Number: EASA.R.008
Type Certificate Holder: AIRBUS HELICOPTERS
Type: AS 350 / EC 130
Model: AS 350 B3
Original STC Number: FAA SR02719SE

Description of Design Change:

Onboard System Cargo Swing Suspension System for AS 350 B3 fitted with Crash Resistant Fuel Tank System (CRFS) as per TCH mod 0720034.

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/ approval.
The requirements for environmental protection and the associated certified noise and/ or emissions levels of the product are unchanged and remain applicable to this certificate/approval without any impact on the noise database.

Associated Technical Documentation:

Master Drawings List No. 155-216-00, Rev 2, dated 30 March 2021
Instructions For continued Airworthiness No. 123-051-00, Rev 00, dated 4 March 2021
Rotorcraft Fight Manual Supplement No. 121-071-00, Rev 00, dated 27 April 2021

See Continuation Sheet(s)

For the European Union Aviation Safety Agency

Cologne, Germany, 13 August 2021

Fabrice LEGAY
Section Manager
Medium & Light Rotorcraft

Task Number: 60078984
ONBOARD SYSTEMS INTERNATIONAL - 302945

or later revisions of the above listed document(s) approved/accepted on behalf of EASA in accordance with the Technical Implementation Procedures of EU/ USA Bilateral Agreement.

Limitations/Conditions:

Prior to installation of this change/repair it must be determined that the interrelationship between this change/repair and any other previously installed change and/ or repair will introduce no adverse effect upon the airworthiness of the product.

- End -